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| **145.A.42 Acceptance of components** |
| **Sostituito sottoparagrafo (a)** |
| **Vecchio Testo** |
| 1. All components shall be classified and appropriately segregated into the following categories:    1. Components which are in a satisfactory condition, released on an EASA Form 1 or equivalent and marked in accordance with Part-21 Subpart Q.    2. Unserviceable components which shall be maintained in accordance with this section.    3. Unsalvageable components which are classified in accordance with 145.A.42(d).    4. Standard parts used on an aircraft, engine, propeller or other aircraft component when specified in the manufacturer's illustrated parts catalogue and/or the maintenance data.    5. Material both raw and consumable used in the course of maintenance when the organisation is satisfied that the material meets the required specification and has appropriate traceability. All material must be accompanied by documentation clearly relating to the particular material and containing a conformity to specification statement plus both the manufacturing and supplier source. |
| **Nuovo Testo** |
| 1. All components shall be classified and appropriately segregated into the following categories:    1. Components which are in a satisfactory condition, released on an EASA Form 1 or equivalent and marked in accordance with **Subpart Q** **of the Annex (**Part-21**)** ~~Subpart Q~~ **to Regulation (EC) No 1702/2003**.    2. Unserviceable components which shall be maintained in accordance with this section.    3. Unsalvageable components which are classified in accordance with **point** 145.A.42(d).    4. Standard parts used on an aircraft, engine, propeller or other aircraft component when specified in the manufacturer’s illustrated parts catalogue and/or the maintenance data.    5. Material both raw and consumable used in the course of maintenance when the organisation is satisfied that the material meets the required specification and has appropriate traceability. All material must be accompanied by documentation clearly relating to the particular material and containing a conformity to specification statement plus both the manufacturing and supplier source.    6. **Components referred to in point 21A.307(c) of the Annex (Part-21) to Regulation (EC) No 1702/2003.** |
| **Aggiunto sottoparagrafo (e)** |
| **Vecchio Testo** |
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| **Nuovo Testo** |
| 1. **Components referred to in point 21A.307(c) of the Annex (Part-21) to Regulation (EC) No 1702/2003 shall only be installed if considered eligible for installation by the aircraft owner in its own aircraft.** |
| **Razionale/Ulteriori Dettagli**: A parte alcune variazioni di tipo editoriale, i due sotto paragrafi sono stati aggiunti per annoverare le parti e le pertinenze di cui al § 21.A.307 tra le possibili tipologie di componenti e per chiarire che esse sono installabili su un aeromobile solo il relativo proprietario, sotto la propria responsabilità, le ha considerate avere titolo all’istallazione.  In relazione al cambiamento del formato della numerazione dei §§ della Parte 21 introdotto con il Regolamento (UE) No. 748/2012, [vedere scheda M.A.302](MA302.docx#CambioNumParaPart21) |
| **Principali implicazioni:** [Vedi **Principali implicazioni** su scheda § M.A.502](MA502_.docx#PrincImplMA502) |
| **Azioni per le parti interessate:** vedi NI su regolamenti (UE) 593/2012 e 748/2012.  [Vedi **Azioni per le parti interessate** su scheda § M.A.502](MA502_.docx#AZSTAKEHOLDMA502) |