PREDEFINED RISK ASSESSMENT



IT-PDRA-07: BVLOS flight close to obstacles over Urban area

(a) Scope

This PDRA is the result of applying the SORA methodology to UAS operations performed in the 'specific' category with the following main attributes:

- 1) Multirotor UA:
 - a. With maximum characteristic dimensions (i.e. maximum distance between rotors) up to 1 m;
 - b. With a Take-Off Mass (including payload) up to 2 Kg.;
 - c. Compliant with requirements defined in ENAC LG 2016/003-NAV Ed. 1 1/6/2016 for harmless UAS.
- 2) operated BVLOS of the remote pilot;
- 3) over populated areas;
- 4) less than 15 meters above the highest obstacle and in any case less than 120 m (400 ft) above the overflown surface; and
- 5) in a volume of airspace where BVLOS is allowed referring to the geo-awareness maps on D-flight portal.

(b) PDRA characterisation and provisions

Characterisation and provisions for this PDRA are summarised in the following table:

PDRA characterisation and provisions					
Item	Requirement	Means of Evidence	Supporting Material		
1. Operational chara	acterisation (scope and limitations)				
Level of human intervention	 No autonomous operations: the remote pilot should have the ability to control the UA, except in case of a lost link. The remote pilot only operates one UA at a time. The remote pilot does not operate from a moving vehicle. Handover between RPSs is not performed. 	Operations	The Operations Manual shall be developed in accordance with the template provided in Appendix D		
UA range limit	1.5. Launch/recovery: VLOS distance from the remote pilot 1.6. In flight: 1.6.1. UA is not operated at more than 2 km from the remote pilot.	─ Manual			
Overflown areas	1.7. Populated areas.				

	1.8. Maximum characteristic dimension (e.g. rotor diameter/area or maximum distance between rotors, wing-span): 1 m	Harmless	
UA limitations	1.9. Take off mass (including payload) up to 2 kg1.10. Compliant with requirements defined in ENAC LG 2016/003-NAV Ed. 11/6/2016 for harmless UAS.	certification	
Flight height limit	1.11. The maximum height of the operational volume is not greater than 15 meters above the highest height of the obstacle(s) and in any case 120 m (400 ft) above the overflown surface.		
	Note: In addition to the vertical limit for the operational volume, an air risk buffer is to be considered (see 'air risk' under point 3 of this table).		
Airspace	1.12. Operated: 1.12.1 in operational volume within the volumes allowed by ATM-09, unless the UAS operator is in receipt of the appropriate permission, close to the obstacle(s) (corresponding to an air risk that can be classified as ARC-a);		
Visibility	N/A		
Others	1.13. The UA should not be used to drop material or carry dangerous goods, except for dropping items in connection with agricultural, horticultural or forestry activities in which the carriage of the items does not contravene any other applicable regulations.		

2. Operational risk classification (according to SORA)					
Final GRC	3	Final ARC	Arc-a	SAIL	II

Item		Requirement	Means of Evidence	Supporting Material		
3. Operational n	3. Operational mitigations					
Operational volume	3.1.	To determine the operational volume, the applicant considers the position-keeping capabilities of the UAS in 4D space (latitude, longitude, height and time). In particular, the accuracy of the navigation solution, the flight technical error of the UAS and the path definition error	Operations Manual	The Operations Manual shall be developed in accordance with the template provided		



		(e.g. map error) and latencies are considered and addressed		
		in this determination.		
	3.3.	, ,		
		procedures are activated immediately.		
	3.4.	A ground risk buffer is established to protect third parties on	Operations Manual	
		the ground.		
		3.4.1. The minimum criterion should be the use of the '1:1	Note: The applicant should evaluate the	
Ground risk		rule' (e.g. if the UA is planned to operate at a height of	area of operations by means of an on-site	
Ground risk		120 m, the ground risk buffer should at least be 120	inspection or appraisal, and should be able	
		m).	to determine that the area is non-	
	3.5.	The operational volume and the ground risk buffer is all	populated. This procedure shall be included	
		contained in a non-populated environment.	in the Operations Manual	
	3.6.			
		allowed by ATM-09, unless the UAS operator is in receipt of		
		the appropriate permission.		
	3.7.	, , , ,		
Air risk	3.8.	• • •	Operations Manual	
		ATM-09		
	3.9.			
		manned aircraft activity should be assessed.		
VOs	N/A	,		
4. Operator pro			I	L
	4.1.			The Operations Manual
		4.1.1.have knowledge of the UAS being used; and		shall be developed in
		4.1.2.develop relevant procedures including at least the		accordance with the
Operator		following as a minimum: operational procedures (e.g.	Operations Manual	template provided
		checklists), maintenance, training, responsibilities,		
		and duties.		
	4.2.		Operations Manual	The Operations Manual
		standards recognised by the competent authority and/or in	Emergency Response Plan (ERP)	and the ERP shall be
UAS operations		accordance with a means of compliance acceptable to that		developed in
		authority.	Notes:	accordance with the
2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	4.3.	•	The adequacy of the contingency and	template provided
		Plan	emergency procedures should be	The second second
	4.4.	The remote crew should be competent and be authorised by	proved through:	
			, , , , , , , , , , , , , , , , , , ,	



	 the UAS operator to carry out the intended operations. 4.5. A list of the remote crew members authorised to carry out UAS operations is established and kept up to date. 4.6. The applicant should have a policy that defines how the remote crew can declare themselves fit to operate before conducting any operation. 	 dedicated flight tests; or simulations, provided that the representativeness of the simulation means is proven for the intended purpose with positive results; Operations Manual should include an up-to-date record of all the relevant qualifications, experience and/or training completed by the remote crew. 	
UAS maintenance	 4.7. The UAS maintenance instructions should be defined by the UAS operator, documented and cover at least the UAS manufacturer's instructions and requirements when applicable. 4.8. The maintenance staff should be competent and should have received an authorisation from the UAS operator to carry out maintenance. 4.9. The maintenance staff should use the UAS maintenance instructions while performing maintenance. 	I AUTHORISEA TO CARRY OUT MAINTENANCE	The Operations Manual shall be developed in accordance with the template provided
External services	4.10. The applicant should ensure that the level of performance for any externally provided service necessary for the safety of	Operations Manual	The Operations Manual shall be developed in

		the flight is adequate for the intended operation.	Note: The applicant should declare that	accordance with the
	4.11	The roles and responsibilities between the applicant and the	this adequate level of performance is	template provided
		external service provider should be defined in the Operations Manual.	achieved.	
5. Provisions for	the p	ersonnel in charge of duties essential to the UAS operation		
	Ref.	LG 2020/001-NAV para 6.4		
6. Technical pro-	visions	S		
General	6.1.	Means to monitor critical parameters for a safe flight should be available, in particular the: 6.1.1.UA position, height or altitude, ground speed or airspeed, attitude and trajectory; 6.1.2.UAS energy status (fuel, battery charge, etc.); and the 6.1.3.status of critical functions and systems; as a minimum, for services based on RF signals (e.g. C2 Link, GNSS, etc.), means should be provided to monitor the adequate performance and trigger an alert if the level becomes too low. The UA should have the performance capability to descend safely from its operating altitude to a 'safe altitude' in less than a minute, or have a descent rate of at least 2.5 m/s (500 fpm).	Flight Manual	
НМІ	6.3.	The UAS information and control interfaces should be clearly and succinctly presented and should not confuse, cause unreasonable fatigue, or contribute to causing any disturbance to the personnel in charge of duties essential to the UAS operation such that this could adversely affect the safety of the operation. The applicant should conduct an evaluation of the UAS considering and addressing human factors to determine whether the HMI is appropriate for the mission.	Flight Manual	
C2 links and communication	6.5. 6.6. 6.7.	The UAS should comply with the appropriate requirements for radio equipment and the use of the RF spectrum. The Primary C2L shall operate in Radio Line of Sight Protection mechanisms against interference should be used, especially if unlicensed bands (e.g. ISM) are used for the C2	Operations Manual	The Operations Manual shall be developed in accordance with the template provided

	Link (mechanisms such as FHSS, technology or frequency de-		
	confliction by procedure).		l l
	6.8. The UAS shall be equipped with a C2 Link Recovery function		•
	in case of loss.		
	6.9. The UAS design should be adequate to ensure that the time		
	required between a command given by the remote pilot and		
	the UA executing it does not exceed 5 seconds.		
Tactical	6.10. Where an electronic means is used to assist the remote pilot	Flight Manual	
mitigation	in being aware of the UA position in relation to potential		
	'airspace intruders', the information is provided with a		
	latency and an update rate for intruder data (e.g. position,		
	speed, altitude, track) that support the decision criteria.		
	6.11. To ensure a safe recovery from a technical issue involving the		
	UAS or an external system supporting the operation, the UAS		
	operator should ensure:		
	6.11.1. that no probable failure of the UAS or any external		
	system supporting the operation should lead to		
	operation outside the operational volume.		
	6.11.2. that it is reasonably expected that a fatality will not	Compliance to all these requirements can	
	occur from any probable failure of the UAS, or any	be ensured by using both:	
	external system supporting the operation.	a Flight Termination system that is	
	6.12. The vertical extension of the operational volume should be	independent and dissimilar from	
	120 m above the surface.	•	
Containment	Note: The term 'probable' needs to be understood in its qualitative	the Primary Control System	
	interpretation, i.e. 'anticipated to occur one or more times during	a geo-fencing function	
	the entire system/operational life of an item.'	F. ideas of each accordance at a starting the	
		Evidence of analyses demonstrating the	
	1 6 13 A design and installation appraisal should be made available	effectiveness of the containment measures	
	and should minimally include:	shall be provide.	
	6.13.1. design and installation features (independence,		
	separation and redundancy);		
	6.13.2. particular risks (e.g. hail, ice, snow, electro-		
	magnetic interference, etc.) relevant to the ConOps.		
	6.14. The following additional provisions should apply if the		
	adjacent area includes an assembly of people or if the		

ENAC

adjacent airspace is classified as ARC-d (in accordance with AMC1 to Article 11 of the UAS Regulation):

- 6.14.1. The probability of leaving the operational volume should be less than 10-4/FH.
- 6.14.2. No single failure of the UAS or any external system supporting the operation should lead to operation outside the ground risk buffer.

Note: The term 'failure' needs to be understood as an occurrence, which affects the operation of a component, part, or element such that it can no longer function as intended. Errors may cause failures but are not considered to be failures. Some structural or mechanical failures may be excluded from the criterion if it can be shown that these mechanical parts were designed according to aviation industry best practices.

6.14.3. SW and AEH whose development error(s) could directly lead to operations outside the ground risk buffer should be developed to an industry standard or methodology recognised as adequate by the competent authority.

Note 1: The proposed additional safety provisions cover both the integrity and assurance levels.

Note 2: The proposed additional safety provisions do not imply a systematic need to develop the SW and AEH according to an industry standard or methodology recognised as adequate by the competent authority. For instance, if the UA design includes an independent engine shutdown function which systematically prevents the UA from exiting the ground risk buffer due to single failures or a SW/AEH error of the flight controls, the intent of provisions 6.16.2 and 6.16.3 could be considered to be met.